

TALAT Lecture 2301

Design of Members

Axial Force

Example 5.3 : Resistance of cross section with radiating outstands

6 pages

Advanced Level

prepared by Torsten Höglund, Royal Institute of Technology, Stockholm

Date of Issue: 1999

© EAA - European Aluminium Association

Example 5.3. Resistance of cross section with radiating outstands

Flange width $b := 80 \cdot \text{mm}$ Length $L := 1200 \cdot \text{mm}$
 Inner circle $c := 18 \cdot \text{mm}$ $f_o := 300 \cdot \text{MPa}$ $\gamma_{MF} = 1.0$ $kN \equiv 1000 \cdot \text{newton}$
 Inner thickness $t_i := 10 \cdot \text{mm}$ $E := 70000 \cdot \text{MPa}$ $G := 27000 \cdot \text{MPa}$ $\text{MPa} \equiv 10^6 \cdot \text{Pa}$
 Outer thickness $t_o := 5 \cdot \text{mm}$

$$k := 1, 4 \dots 16 \quad \phi_{k-1} := \frac{\pi}{4} \cdot \frac{k-1}{3} - \frac{\pi}{4} \quad \phi_k := \phi_{k-1}$$

$$\phi_{k+1} := \frac{\pi}{4} \cdot \frac{k+2}{3} - \frac{\pi}{4} \quad \phi_{18} := \phi_{17}$$

Nodes no.,
co-ordinates,
thickness

$$i := 0 \dots 16 \quad y_{k-1} := b \cdot \cos(\phi_{k-1}) \quad z_{k-1} := b \cdot \sin(\phi_{k-1})$$

$$y_k := c \cdot \cos(\phi_k) \quad z_k := c \cdot \sin(\phi_k)$$

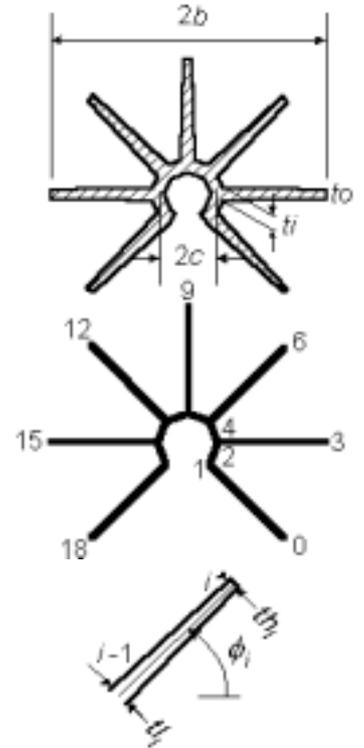
$$y_{k+1} := c \cdot \cos(\phi_{k+1}) \quad z_{k+1} := c \cdot \sin(\phi_{k+1})$$

$$y_{18} := y_{12}$$

$$z_{18} := z_0$$

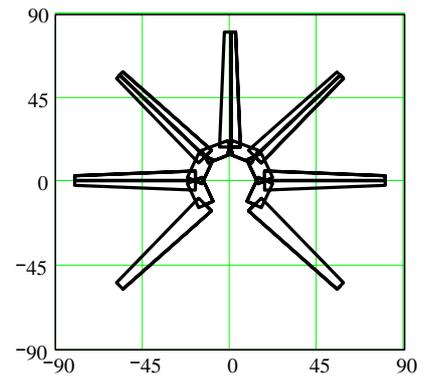
$$t_{k-1} := o \quad th_{k-1} := o \quad t_k := t_o \quad th_k := t_i$$

$$t_{k+1} := t_i \cdot 0.8 \quad th_{k+1} := t_{k+1} \quad t_{18} := t_i \quad th_{18} := t_o$$



$$i = \phi_i = \frac{y_i}{\text{mm}} = \frac{z_i}{\text{mm}} = \frac{t_i}{\text{mm}} = \frac{th_i}{\text{mm}} =$$

1	-0.785	12.728	-12.728	5	10
2	0	18	0	8	8
3	0	80	0	0	0
4	0	18	0	5	10
5	0.785	12.728	12.728	8	8
6	0.785	56.569	56.569	0	0
7	0.785	12.728	12.728	5	10
8	1.571	0	18	8	8
9	1.571	0	80	0	0
10	1.571	0	18	5	10
11	2.356	-12.728	12.728	8	8
12	2.356	-56.569	56.569	0	0
13	2.356	-12.728	12.728	5	10
14	3.142	-18	0	8	8
15	3.142	-80	0	0	0
16	3.142	-18	0	5	10
17	3.927	-12.728	-12.728	8	8
18	3.927	-56.569	-56.569	10	5



Nodes $i := 1 \dots \text{rows}(y) - 1$

Cross-section constants, varying thickness

Sectorial
co-ordinates

$$\omega_0 := 0 \cdot \text{mm}^2$$

$$\omega_i := y_{i-1} \cdot z_i - y_i \cdot z_{i-1}$$

$$\omega_i := \omega_{i-1} + \omega_0$$

$$Dt_i := th_i - tl_i$$

$$tl_i := tl_i + \frac{Dt_i}{2}$$

$$t2_i := \frac{tl_i}{2} + \frac{Dt_i}{3}$$

$$t3_i := \frac{tl_i}{3} + \frac{Dt_i}{4}$$

Length of
elements

$$l_i := \sqrt{(y_i - y_{i-1})^2 + (z_i - z_{i-1})^2}$$

Area

$$A := \sum_{i=1}^{\text{rows}(y)-1} \frac{tl_i + th_i}{2} \cdot l_i$$

$$A = 3.916 \cdot 10^3 \cdot \text{mm}^2$$

First moments
of area

$$S_z := \sum_{i=1}^{\text{rows}(y)-1} [y_{i-1} \cdot tl_i + (y_i - y_{i-1}) \cdot t2_i] \cdot l_i$$

$$S_z = 1.186 \cdot 10^{11} \cdot \text{mm}^3$$

$$S_y := \sum_{i=1}^{\text{rows}(y)-1} [z_{i-1} \cdot tl_i + (z_i - z_{i-1}) \cdot t2_i] \cdot l_i$$

$$S_y = 2.457 \cdot 10^4 \cdot \text{mm}^3$$

$$S_\omega := \sum_{i=1}^{\text{rows}(y)-1} [\omega_{i-1} \cdot tl_i + (\omega_i - \omega_{i-1}) \cdot t2_i] \cdot l_i$$

$$S_\omega = 2.692 \cdot 10^6 \cdot \text{mm}^4$$

Second
moments
of area

$$I_z := \sum_{i=1}^{\text{rows}(y)-1} [(y_{i-1})^2 \cdot tl_i + 2 \cdot y_{i-1} \cdot (y_i - y_{i-1}) \cdot t2_i + (y_i - y_{i-1})^2 \cdot t3_i] \cdot l_i$$

$$I_z = 4.551 \cdot 10^6 \cdot \text{mm}^4$$

$$I_y := \sum_{i=1}^{\text{rows}(y)-1} [(z_{i-1})^2 \cdot tl_i + 2 \cdot z_{i-1} \cdot (z_i - z_{i-1}) \cdot t2_i + (z_i - z_{i-1})^2 \cdot t3_i] \cdot l_i$$

$$I_y = 3.402 \cdot 10^6 \cdot \text{mm}^4$$

$$I_\omega := \sum_{i=1}^{\text{rows}(y)-1} [(\omega_{i-1})^2 \cdot tl_i + 2 \cdot \omega_{i-1} \cdot (\omega_i - \omega_{i-1}) \cdot t2_i + (\omega_i - \omega_{i-1})^2 \cdot t3_i] \cdot l_i$$

Mixed

$$I_\omega = 2.638 \cdot 10^9 \cdot \text{mm}^6$$

$$I_{yz} := \sum_{i=1}^{\text{rows}(y)-1} [(y_{i-1}) \cdot (z_{i-1}) \cdot tl_i + [y_{i-1} \cdot (z_i - z_{i-1}) + z_{i-1} \cdot (y_i - y_{i-1})] \cdot t2_i + (y_i - y_{i-1}) \cdot (z_i - z_{i-1}) \cdot t3_i] \cdot l_i$$

$$I_{yz} = -1.059 \cdot 10^{10} \cdot \text{mm}^4$$

$$I_{y\omega} := \sum_{i=1}^{\text{rows}(y)-1} [(y_{i-1}) \cdot (\omega_{i-1}) \cdot tl_i + [y_{i-1} \cdot (\omega_i - \omega_{i-1}) + \omega_{i-1} \cdot (y_i - y_{i-1})] \cdot t2_i + (y_i - y_{i-1}) \cdot (\omega_i - \omega_{i-1}) \cdot t3_i] \cdot l_i$$

$$I_{y\omega} = -5.018 \cdot 10^7 \cdot \text{mm}^5$$

$$I_{z\omega} := \sum_{i=1}^{\text{rows}(y)-1} \left[(z_{i-1}) \cdot (\omega_{i-1}) \cdot t l_i + [z_{i-1} \cdot (\omega_i - \omega_{i-1}) + \omega_{i-1} \cdot (z_i - z_{i-1})] \cdot t 2_i + (z_i - z_{i-1}) \cdot (\omega_i - \omega_{i-1}) \cdot t 3_i \right] \cdot l_i$$

$$I_{z\omega} = 1.689 \cdot 10^7 \cdot \text{mm}^5$$

$$I a_i := \overline{\left[(t l_i + t h_i) \cdot \left[(t l_i)^2 + (t h_i)^2 \right] \cdot \frac{l_i}{48} \right]}$$

Influence of thickness

$$I_{yz} := I_{yz} + \sum_{i=1}^{\text{rows}(y)-1} \frac{I a_i \cdot (y_i - y_{i-1}) \cdot (z_i - z_{i-1})}{(l_i)^2}$$

$$I_{yz} = -1.061 \cdot 10^{-10} \cdot \text{mm}^4$$

$$I_z := I_z + \sum_{i=1}^{\text{rows}(y)-1} \frac{I a_i \cdot (z_i - z_{i-1})^2}{(l_i)^2}$$

$$I_z = 4.56 \cdot 10^6 \cdot \text{mm}^4$$

$$I_y := I_y + \sum_{i=1}^{\text{rows}(y)-1} \frac{I a_i \cdot (y_i - y_{i-1})^2}{(l_i)^2}$$

$$I_y = 3.413 \cdot 10^6 \cdot \text{mm}^4$$

$$I_t := \sum_{i=1}^{\text{rows}(y)-1} (t l_i + t h_i) \cdot \left[(t l_i)^2 + (t h_i)^2 \right] \cdot \frac{l_i}{12} \cdot 1.05$$

$$I_t = 8.602 \cdot 10^4 \cdot \text{mm}^4$$

$$z_{gc} := \frac{S_y}{A} \quad z_{gc} = 6.274 \cdot \text{mm} \quad I_y := I_y - A \cdot z_{gc}^2 \quad I_y = 3.259 \cdot 10^6 \cdot \text{mm}^4$$

$$y_{gc} := \frac{S_z}{A} \quad y_{gc} = 3.028 \cdot 10^{-15} \cdot \text{mm} \quad I_z := I_z - A \cdot y_{gc}^2 \quad I_z = 4.56 \cdot 10^6 \cdot \text{mm}^4$$

$$I_{yz} := I_{yz} - \frac{S_y \cdot S_z}{A} \quad I_{yz} = -1.805 \cdot 10^{-10} \cdot \text{mm}^4 \quad I_{y\omega} := I_{y\omega} - \frac{S_z \cdot S_\omega}{A} \quad I_{y\omega} = -5.018 \cdot 10^7 \cdot \text{mm}^5$$

$$I_\omega := I_\omega - \frac{S_\omega^2}{A} \quad I_\omega = 7.875 \cdot 10^8 \cdot \text{mm}^6 \quad I_{z\omega} := I_{z\omega} - \frac{S_y \cdot S_\omega}{A} \quad I_{z\omega} = -3.309 \cdot 10^{-9} \cdot \text{mm}^5$$

Principal axis

$$\alpha := \text{if} \left[|I_z - I_y| < (I_y + I_z) \cdot 10^{-9}, 0, \frac{1}{2} \cdot \text{atan} \left(\frac{2 \cdot I_{yz}}{I_z - I_y} \right) \right] \quad \alpha \cdot \frac{180}{\pi} = -7.944 \cdot 10^{-15}$$

$$I_\xi := \frac{1}{2} \cdot \left[I_y + I_z + \sqrt{(I_z - I_y)^2 + 4 \cdot I_{yz}^2} \right] \quad I_\xi = 4.56 \cdot 10^6 \cdot \text{mm}^4$$

$$I_\eta := \frac{1}{2} \cdot \left[I_y + I_z - \sqrt{(I_z - I_y)^2 + 4 \cdot I_{yz}^2} \right] \quad I_\eta = 3.259 \cdot 10^6 \cdot \text{mm}^4$$

Shear centre

$$y_{sc} := \frac{I_z \omega \cdot I_z - I_{yz} \cdot I_{yz}}{I_y \cdot I_z - I_{yz}^2} \quad z_{sc} := \frac{-I_{yz} \omega \cdot I_y + I_z \omega \cdot I_{yz}}{I_y \cdot I_z - I_{yz}^2}$$

$$y_{sc} = -1.625 \cdot 10^{-15} \text{ mm}$$

$$z_{sc} = 11.005 \text{ mm}$$

Warping constant

$$I_w := I_\omega + z_{sc} \cdot I_{y\omega} - y_{sc} \cdot I_{z\omega}$$

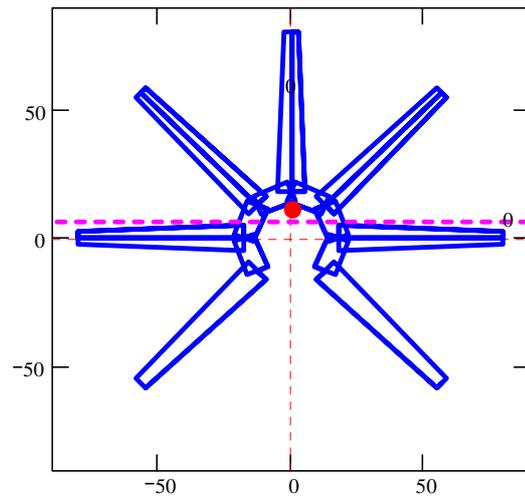
$$I_w = 2.353 \cdot 10^8 \text{ mm}^6$$

Polar radius gyration

$$i_p := \sqrt{\frac{I_y + I_z}{A} + (y_{sc} - y_{gc})^2 + (z_{sc} - z_{gc})^2}$$

$$i_p = 44.932 \text{ mm}$$

Dashed line = gravity centre
Point = shear centre



Axial force resistance, flexural-torsional buckling

Buckling length $l := L$ $l = 1200 \text{ mm}$

Reference buckling loads

$$N_{Ey} := \frac{\pi^2 \cdot E \cdot I_y}{l^2} \quad N_{Ez} := \frac{\pi^2 \cdot E \cdot I_z}{l^2} \quad N_T := \left(G \cdot I_t + \frac{\pi^2 \cdot E \cdot I_\omega}{l^2} \right) \cdot \frac{1}{i_p^2}$$

$$N_{Ey} = 1.56 \cdot 10^3 \text{ kN} \quad N_{Ez} = 2.19 \cdot 10^3 \text{ kN} \quad N_T = 1.34 \cdot 10^3 \text{ kN} \quad \text{Guess: } N := 0.2 \cdot N_{Ez}$$

Given

$$(N_{Ey} - N) \cdot (N_{Ez} - N) \cdot (N_T - N) \cdot i_p^2 - (z_{sc} - z_{gc})^2 \cdot N^2 \cdot (N_{Ey} - N) - (y_{sc} - y_{gc})^2 \cdot N^2 \cdot (N_{Ez} - N) = 0$$

Solution

$$N_{cr} := \text{minerr}(N) \quad N_{cr} = 1.32 \cdot 10^3 \text{ kN}$$

Cross-section entirely of radiating outstands, no welding, unsymmetrical section

Table 5.5 and 5.8 $\alpha := 0.2$ $\lambda_{I1} := 0.6$ $k_{I2} := 1$

$$\left(\lambda_c = \lambda_{bar}\right) \quad \lambda_c := \sqrt{\frac{f_o \cdot A}{N_{cr}}} \quad \lambda_{\bar{c}} = 0.945$$

$$(5.37) \quad \phi := 0.5 \cdot \left[1 + \alpha \cdot (\lambda_c - \lambda_{I1}) + \lambda_c^2 \right] \chi := \frac{1}{\phi + \sqrt{\phi^2 - \lambda_c^2}} \quad \phi = 0.981 \quad \chi = 0.804$$

$$\text{Table 5.5} \quad \psi := \frac{|\min(z - z_{gc})| - |\max(z - z_{gc})|}{|\min(z - z_{gc})| + |\max(z - z_{gc})|} \quad \min(z - z_{gc}) = -62.8 \cdot \text{mm} \quad \psi = -0.08$$

$$\text{Table 5.5} \quad k_{I1} := 1 - 2.4 \cdot \psi^2 \cdot \frac{\lambda_c^2}{(1 + \lambda_c^2) \cdot (1 + \lambda_{I1}^2)^2} \quad \max(z - z_{gc}) = 73.7 \cdot \text{mm} \quad k_{I1} = 0.998$$

$$5.8.3 (1) \quad N_{b.Rd} := \chi \cdot k_{I1} \cdot k_{I2} \cdot \frac{f_o}{\gamma_{M1}} \cdot A \quad N_{b.Rd} = 942.32 \cdot \text{kN}$$