

TALAT Lecture 2301

Design of Members

Axial force and bending moment

Example 9.3 : Beam-column with eccentric load

6 pages

Advanced Level

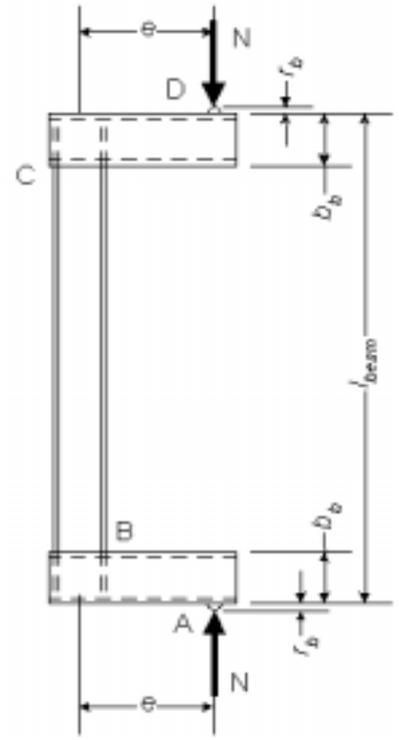
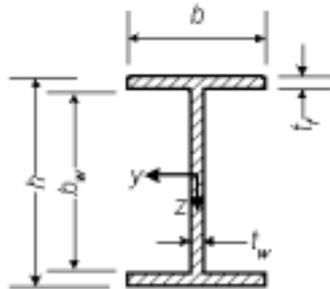
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Example 9.3. Beam-column with eccentric load

The beam-column is subject to an eccentric axial force with same eccentricity at both ends. The beam is prevented from warping at the ends by rigid rectangular hollow beams. These beams are simply supported at the load points and D. The structure is a test beam.



Dimensions and material properties

Section height:	$h := 100.5 \cdot \text{mm}$
Flange depth:	$b := 50.2 \cdot \text{mm}$
Web thickness:	$t_w := 5.07 \cdot \text{mm}$
Flange thickness:	$t_f := 5.06 \cdot \text{mm}$
Overall length:	$l_{\text{beam}} := 800 \cdot \text{mm}$
Cross beams:	$b_b := 140 \cdot \text{mm}$
Support	$r_b := 30 \cdot \text{mm}$
Eccentricity	$exc := 300 \cdot \text{mm}$

Alloy: **EN AW-6082 T6** $heat_treated := 1$ (if heat-treated then 1 else 0)

Note: Values from tests! $f_{0.2} := 300 \cdot \text{MPa}$

[1] (5.4), (5.5) $f_o := f_{0.2}$ $E := 70000 \cdot \text{MPa}$ $G := 27000 \cdot \text{MPa}$
 [1] (5.6)

Partial safety factors: $\gamma_{M1} = 1.10$ $\gamma_{M2} = 1.25$
 Inner radius: $r := 0 \cdot \text{mm}$
 Web height: $b_w := h - 2 \cdot t_f - 2 \cdot r$ $b_w = 90 \cdot \text{mm}$

Moment and force

Axial force (compression) $N_{Ed} := 24.8 \cdot \text{kN}$
 Bending moment at the ends $M_{y,Ed} := N_{Ed} \cdot exc$ $M_{y,Ed} = 7.44 \cdot \text{kNm}$
 S.I. units: $\text{kN} \equiv 1000 \cdot \text{newton}$ $\text{kNm} \equiv \text{kN} \cdot \text{m}$ $\text{MPa} \equiv 1000000 \cdot \text{Pa}$

Cross section class

	Axial force	Flange	Web
[1] 5.4.3 and [1] 5.4.4	$\varepsilon := \frac{250 \cdot \text{MPa}}{f_o}$	$\beta_f = \frac{b - t_w}{2 \cdot t_f}$	$\beta_w = \frac{b_w}{t_w}$
	Heat treated material	$\beta_{1f} = 3 \cdot \varepsilon$ $\beta_{2f} = 4.5 \cdot \varepsilon$ $\beta_{3f} = 6 \cdot \varepsilon$	$\beta_{1w} = 11 \cdot \varepsilon$ $\beta_{2w} = 16 \cdot \varepsilon$ $\beta_{3w} = 22 \cdot \varepsilon$
		$\beta_{1f} = 2.5$ $\beta_{2f} = 3.75$ $\beta_{3f} = 5$	$\beta_{1w} = 17.8$ $\beta_{2w} = 13.3$ $\beta_{3w} = 18.3$
		$class_f := \text{if}(\beta_f > \beta_{1f}, \text{if}(\beta_f > \beta_{2f}, \text{if}(\beta_f > \beta_{3f}, 4, 3), 2), 1)$	$class_w = 3$ (flange)
		$class_w := \text{if}(\beta_w > \beta_{1w}, \text{if}(\beta_w > \beta_{2w}, \text{if}(\beta_w > \beta_{3w}, 4, 3), 2), 1)$	$class_w = 3$ (web)
		$class_c := \text{if}(class_f < class_w, class_w, class_f)$	$class_c = 3$ (compression)

Bending

As web class > class_c and flange class the same, $class_y := 3$ for y-y-axis bending and
 $class_z := 3$ for z-z-axis bending

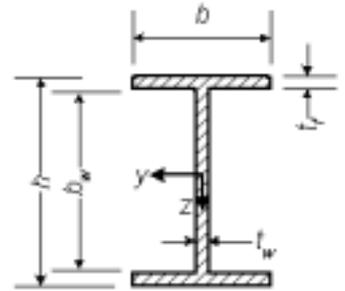
Design resistance, y-y-axis bending

[1] 5.6.1 Elastic modulus of the gross cross section W_{el} :

$$A_{gr} := 2 \cdot b \cdot t_f + (h - 2 \cdot t_f) \cdot t_w \quad A_{gr} = 966.251 \cdot \text{mm}^2$$

$$I_g := \frac{1}{12} \cdot [b \cdot h^3 - (b - t_w) \cdot (h - 2 \cdot t_f)^3] \quad I_g = 1.47 \cdot 10^6 \cdot \text{mm}^4$$

$$W_{el} := \frac{I_g \cdot 2}{h} \quad W_{el} = 2.925 \cdot 10^4 \cdot \text{mm}^3$$



[1] Tab. 5.3

[1] (5.15) Shape factor α for class 3 cross-section, say

$$\alpha_y = 1$$

[1] (5.14) Design moment of resistance of the cross section $M_{c,Rd}$

$$M_{y,Rd} := \frac{f_o \cdot \alpha_y \cdot W_{el}}{\gamma_{MI}}$$

$$M_{y,Rd} = 8 \cdot \text{kNm}$$

Axial force resistance, y-y buckling

[1] 5.8.4 Cross section area of gross cross section A_{gr}

$$A_{gr} := b \cdot h - (b - t_w) \cdot (h - 2 \cdot t_f)$$

$$A_{gr} = 966.3 \cdot \text{mm}^2$$

Cross section area of class 3 cross section

$$A_{eff} := A_{gr}$$

Effective cross section factor

$$\eta := \frac{A_{eff}}{A_{gr}}$$

$$\eta = 1$$

Second moment of area of gross cross section:

$$I_y := \frac{2}{12} \cdot b \cdot t_f^3 + 2 \cdot b \cdot t_f \cdot \left(\frac{h - t_f}{2}\right)^2 + \frac{1}{12} \cdot (h - 2 \cdot t_f)^3 \cdot t_w$$

$$I_y = 1.47 \cdot 10^6 \cdot \text{mm}^4$$

[1] Table 5.7	Effective buckling length	$l_{yc} := l_{beam} + 2 \cdot r_b$	$l_{yc} = 860 \cdot mm$
	Buckling load	$N_{cr} := \frac{\pi^2 \cdot E \cdot I_y}{l_{yc}^2}$	$N_{cr} = 1.373 \cdot 10^3 \cdot kN$
[1] 5.8.4.1	Slenderness parameter	$\lambda_{\bar{y}} := \sqrt{\frac{A_{gr} \cdot \eta \cdot f_o}{N_{cr}}}$	$\lambda_{\bar{y}} = 0.459$
[1] Table 5.6	$\alpha := if(heat_treated=1, 0.2, 0.32)$		$\alpha = 0.2$
	$\lambda_{\bar{o}} := if(heat_treated=1, 0.1, 0)$		$\lambda_{\bar{o}} = 0.1$
	$\phi := 0.5 \cdot \left[1 + \alpha \cdot (\lambda_{\bar{y}} - \lambda_{\bar{o}}) + \lambda_{\bar{y}}^2 \right]$		$\phi = 0.642$
	$\chi_{\bar{y}} := \frac{1}{\phi + \sqrt{\phi^2 - \lambda_{\bar{y}}^2}}$		$\chi_{\bar{y}} = 0.918$
[1] Table 5.5	Symmetric profile		$k_1 := 1$
[1] Table 5.5	No longitudinal welds		$k_2 := 1$
	Axial force resistance	$N_{y,Rd} := \chi_{\bar{y}} \cdot \eta \cdot k_1 \cdot k_2 \cdot \frac{f_o}{\gamma_{MI}} \cdot A_{gr}$	$N_{y,Rd} = 241.9 \cdot kN$

Axial force resistance, z-z axis buckling

	Effective buckling length	$l_{zc} := l_{beam} + r_b$	$l_{zc} = 0.83 \cdot m$
	Buckling load	$I_z := \frac{2 \cdot t \cdot f \cdot b^3}{12}$ $N_{cr} := \frac{\pi^2 \cdot E \cdot I_z}{l_{zc}^2}$	$N_{cr} = 107 \cdot kN$
[1] 5.8.4.1	Slenderness parameter	$\lambda_{\bar{z}} := \sqrt{\frac{A_{gr} \cdot \eta \cdot f_o}{N_{cr}}}$	$\lambda_{\bar{z}} = 1.646$
[1] 5.8.4.1	$\phi := 0.5 \cdot \left[1 + \alpha \cdot (\lambda_{\bar{z}} - \lambda_{\bar{o}}) + \lambda_{\bar{z}}^2 \right]$		$\phi = 2.009$
	$\chi_{\bar{z}} := \frac{1}{\phi + \sqrt{\phi^2 - \lambda_{\bar{z}}^2}}$		$\chi_{\bar{z}} = 0.316$
[1] 5.8.4.1	Axial load resistance	$N_{z,Rd} := \chi_{\bar{z}} \cdot \eta \cdot k_1 \cdot k_2 \cdot \frac{f_o}{\gamma_{MI}} \cdot A_{gr}$	$N_{z,Rd} = 83.352 \cdot kN$
	and without column buckling	$N_{Rd} := \eta \cdot \frac{f_o}{\gamma_{MI}} \cdot A_{gr}$	$N_{Rd} = 263.5 \cdot kN$

Flexural buckling of beam-column

[1] (5.51) $\omega_{\bar{\theta}} = 1$ $\omega_x := 1$

Exponents in interaction formulae

[1] (5.42c) $\xi_0 := \alpha_y^2$ $\xi_0 := \text{if}(\xi_0 < 1, 1, \xi_0)$ $\xi_{\bar{\theta}} = 1$

[1] 5.9.4.2 $\xi_{y\bar{c}} = \xi_0 \cdot \chi_y$ $\xi_{y\bar{c}} := \text{if}(\xi_{y\bar{c}} < 0.8, 0.8, \xi_{y\bar{c}})$ $\xi_{y\bar{c}} = 0.918$

Flexural buckling check

$N_{Ed} = 24.8 \cdot kN$

$M_{y,Ed} = 7.44 \cdot kNm$

[1] 5.4.4
$$U_y := \left(\frac{N_{Ed}}{\chi_{\bar{\theta}} \cdot \omega_x \cdot N_{Rd}} \right)^{\xi_{y\bar{c}}} + \frac{M_{y,Ed}}{\omega_{\bar{\theta}} \cdot M_{y,Rd}}$$
 $U_y = 1.056$

Lateral-torsional buckling

[1] Annex J Figure J.2 Warping constant:
$$I_w := \frac{(h - t_f)^2 \cdot I_z}{4}$$
 $I_w = 2.429 \cdot 10^8 \cdot mm^6$

Torsional constant:
$$I_t := \frac{2 \cdot b \cdot t_f^3 + h \cdot t_w^3}{3}$$
 $I_t = 8.702 \cdot 10^3 \cdot mm^4$

Lateral buckling length $L := l_{beam} - 2 \cdot b_b$ $W_y := W_{el}$ $W_y = 2.925 \cdot 10^4 \cdot mm^3$

Constant bending moment, warping prevented at the ends

[1] H.1.2(2) C_1, k and k_w constants $C_1 := 1$ $k := 1$ $k_w := 0.5$

[1] H.1.3(2)
$$M_{cr} := \frac{C_1 \cdot \pi^2 \cdot E \cdot I_z}{(k \cdot L)^2} \cdot \sqrt{\left(\frac{k}{k_w} \right)^2 \cdot \frac{I_w}{I_z} + \frac{(k \cdot L)^2 \cdot G \cdot I_t}{\pi^2 \cdot E \cdot I_z}}$$
 $L = 520 \cdot mm$ $M_{cr} = 27.219 \cdot kNm$

[1] 5.6.6.3(3)
$$\lambda_{LT} := \sqrt{\frac{\alpha_y \cdot W_y \cdot f_o}{M_{cr}}}$$
 $\lambda_{LT} = 0.568$

[1] 5.6.6.3(2) $\alpha_{LT} := \text{if}(\text{class}_z > 2, 0.2, 0.1)$ $\alpha_{LT} = 0.2$

$\lambda_{OLT} := \text{if}(\text{class}_z > 2, 0.4, 0.6)$ $\lambda_{OLT} = 0.4$

[1] 5.6.6.3(1)
$$\phi_{LT} := 0.5 \cdot \left[1 + \alpha_{LT} \cdot (\lambda_{LT} - \lambda_{OLT}) + \lambda_{LT}^2 \right]$$
 $\phi_{LT} = 0.678$

$$\chi_{LT} := \frac{1}{\phi_{LT} + \sqrt{\phi_{LT}^2 - \lambda_{LT}^2}}$$
 $\chi_{LT} = 0.954$

Lateral-torsional buckling of beam-column

[1] (5.50) $\omega_{\bar{x}} = 1$ $\omega_{xLT} = 1$ $\omega_0 = 1$

[1] 5.9.4.3 Exponents in interaction expression (simply) $\eta_c = 0.8$ $\gamma_c := 1$ $\xi_{zc} := 0.8$

[1] (5.43)
$$U_z := \left[\left(\frac{N_{Ed}}{\chi_{z\omega_x} N_{Rd}} \right)^{\eta_c} + \left(\frac{M_{y,Ed}}{\chi_{LT\omega_x} M_{y,Rd}} \right)^{\gamma_c} + \left(\frac{M_{z,Ed}}{\omega_0 M_{z,Rd}} \right)^{\xi_{zc}} \right]$$

Utilisation, lateral-torsional buckling

$U_z = 1.357$

Utilisation, flexural buckling

$U_y = 1.056$

Comment: $U_z = 1.357$ mean that the failure load at the test exceeded the characteristic strength with 35.7